

Autonomous parachute

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En virtuel le mardi 3 novembre 2020

<https://youtu.be/eqOGpQ0qoel>



Reach a target

System: $\dot{\mathbf{x}} = \mathbf{f}(\mathbf{x}, \mathbf{w}, \mathbf{u})$

Sensors: $\mathbf{y} = \mathbf{g}(\mathbf{x})$

Input: \mathbf{u}

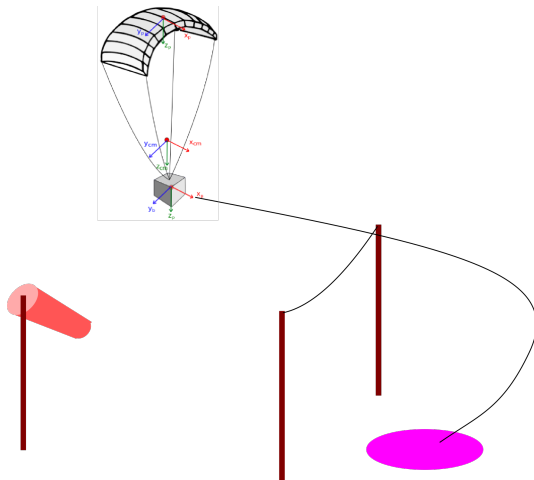
Perturbation: \mathbf{w}

Controller: $\mathbf{u} = \mathbf{r}(\mathbf{y})$

Target: \mathbb{T}

Obstacles: \mathbb{O}

Initial set: \mathbb{X}_0





Experiment with Eole [1]

youtu.be/L04n5RVhgAc

Model

δ_S
 δ_α

$$\begin{pmatrix} \dot{x} \\ \dot{y} \\ \dot{z} \end{pmatrix} = \mathbf{v}_{\text{out}} = \mathbf{H}_B^T \cdot \mathbf{v}_{\text{in}} + \mathbf{v}_w$$

$$\begin{pmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{pmatrix} = \begin{pmatrix} 1 & \tan(\theta)\sin(\phi) & \tan(\theta)\cos(\phi) \\ 0 & \cos(\phi) & -\sin(\phi) \\ 0 & \frac{\sin(\phi)}{\cos(\theta)} & \frac{\cos(\phi)}{\cos(\theta)} \end{pmatrix} \cdot \boldsymbol{\omega}$$

$$\begin{aligned} (m_L + M_F) \ddot{\mathbf{v}}_{\text{in}} &= \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{L\alpha} + C_{L\alpha} \alpha_{\text{ref}}] \begin{pmatrix} \mathbf{v}_{\text{ref}} \\ 0 \\ -\mathbf{v}_{\text{ref}} \end{pmatrix} - \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{D\alpha} + C_{D\alpha} \alpha_{\text{ref}}^2] \mathbf{v}_{\text{ref}} \\ &+ \begin{pmatrix} (C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}}) \text{sign}(\delta_k) & C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}} \\ -C_{D\alpha} \mathbf{v}_{\text{ref}} \text{sign}(\delta_k) & -C_{D\alpha} \mathbf{v}_{\text{ref}} \\ (-C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}}) \text{sign}(\delta_k) & -C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}} \end{pmatrix} \begin{pmatrix} \delta_k \\ \delta_k \end{pmatrix} \\ &- \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{D\alpha} + C_{D\alpha} \alpha_{\text{ref}}^2] \mathbf{v}_3 - \mathbf{Ad}(\boldsymbol{\omega}) \cdot M_F \cdot \mathbf{v}_p + m g \begin{pmatrix} -\sin(\theta) \\ \sin(\phi)\cos(\theta) \\ \cos(\phi)\cos(\theta) \end{pmatrix} \\ &- m \mathbf{Ad}(\boldsymbol{\omega}) \cdot \mathbf{v}_{\text{in}} \end{aligned}$$

$$\begin{aligned} (\mathbf{J} + \mathbf{I}_F) \dot{\boldsymbol{\omega}} &= \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\|^2 \begin{pmatrix} C_{\alpha} b^2 \frac{1}{2|\mathbf{v}_{\text{ref}}|} + C_{L\alpha} b \phi \\ C_{m\alpha} c^2 \frac{1}{2|\mathbf{v}_{\text{ref}}|} + C_{m\alpha} c + C_{m\alpha} \alpha_{\text{ref}} \\ C_{m\alpha} b^2 \frac{1}{2|\mathbf{v}_{\text{ref}}|} \end{pmatrix} + \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\|^2 \begin{pmatrix} C_{L\alpha} \frac{1}{2} & 0 \\ 0 & 0 \\ C_{m\alpha} \frac{1}{2} & 0 \end{pmatrix} \begin{pmatrix} \delta_k \\ \delta_k \end{pmatrix} \\ &- \mathbf{Ad}(\boldsymbol{\omega}) \mathbf{I}_F \cdot \boldsymbol{\omega} - \mathbf{Ad}(\mathbf{v}_p) \cdot M_F \cdot \mathbf{v}_p + \mathbf{Ad}(\mathbf{X}_{\text{ref}}) \left[\frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{L\alpha} + C_{L\alpha} \alpha_{\text{ref}}] \begin{pmatrix} \mathbf{v}_{\text{ref}} \\ 0 \\ -\mathbf{v}_{\text{ref}} \end{pmatrix} \right. \\ &- \frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{D\alpha} + C_{D\alpha} \alpha_{\text{ref}}^2] \mathbf{v}_{\text{ref}} + \begin{pmatrix} (C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}}) \text{sign}(\delta_k) & C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}} \\ -C_{D\alpha} \mathbf{v}_{\text{ref}} \text{sign}(\delta_k) & -C_{D\alpha} \mathbf{v}_{\text{ref}} \\ (-C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}}) \text{sign}(\delta_k) & -C_{L\alpha} \mathbf{v}_{\text{ref}} - C_{D\alpha} \mathbf{v}_{\text{ref}} \end{pmatrix} \begin{pmatrix} \delta_k \\ \delta_k \end{pmatrix} \\ &- M_F \ddot{\mathbf{v}}_{\text{in}} - \mathbf{Ad}(\boldsymbol{\omega}) \cdot M_F \cdot \mathbf{v}_p \left. \right] + \mathbf{Ad}(\mathbf{X}_{\text{ref}}) \left[-\frac{1}{2} \rho S_{\text{ref}} \|\mathbf{v}_{\text{ref}}\| [C_{D\alpha} + C_{D\alpha} \alpha_{\text{ref}}^2] \mathbf{v}_3 \right] \\ &- \mathbf{Ad}(\mathbf{v}_p) \cdot M_F \cdot \mathbf{v}_p - \mathbf{Ad}(\boldsymbol{\omega}) \cdot (\mathbf{J} + \mathbf{I}_F) \cdot \boldsymbol{\omega} \end{aligned}$$

Dubins car

$$\begin{cases} \dot{x}_1 = \cos x_3 + w_1 \\ \dot{x}_2 = \sin x_3 + w_2 \\ \dot{x}_3 = u \end{cases}$$

Controller

Error we want to cancel [2][3]:

$$e = x_3 + a \tan x_2.$$

Thus

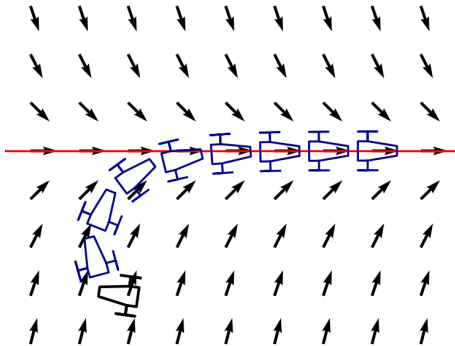
$$\dot{e} = \dot{x}_3 + \frac{\dot{x}_2}{1+x_2^2} = u + \frac{\sin x_3}{1+x_2^2}.$$

We want

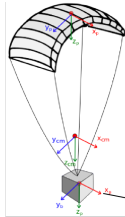
$$\dot{e} + e = 0.$$

We get:

$$u = -x_3 - a \tan x_2 - \frac{\sin x_3}{1+x_2^2}$$



Next step



How to guarantee that if the parachute starts in the region \mathbb{X}_0 , it will reach the target \mathbb{T} for all perturbation $\mathbf{w} \in \mathbb{W}$



Kevin Bedin.

Réalisation d'un parachute autonome.

Projet de fin d'études, ENSTA-Bretagne, Brest, France, 2020.



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Tight Slalom Control for Sailboat Robots.

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L. Jaulin.

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